# National Transportation Safety Board Washington, DC 20594

#### **Brief of Accident**

### Adopted 09/13/2005

LAX02FA266

File No. 18288 08/28/2002 Phoenix, AZ Aircraft Reg No. N635AW Time (Local): 18:44 MST Make/Model: Airbus Industrie / A320-231 Fatal Serious Minor/None Engine Make/Model: International Aero Engines / V2500-A1 Crew 0 5 0 Aircraft Damage: Substantial Pass 0 153 1 Number of Engines: 2 Operating Certificate(s): Flag Carrier/Domestic Name of Carrier: AMERICA WEST AIRLINES Type of Flight Operation: Scheduled; Domestic; Passenger Only Reg. Flight Conducted Under: Part 121: Air Carrier Last Depart, Point: HOUSTON, TX Condition of Light: Day Destination: Same as Accident/Incident Location Weather Info Src: Weather Observation Facility Airport Proximity: On Airport/Airstrip Basic Weather: Visual Conditions Airport Name: Phoenix Sky Harbor Intl. Lowest Ceiling: 13000 Ft. AGL, Broken Visibility: 10.00 SM Runway Identification: 08 Runway Length/Width (Ft): 11490 / 150 Wind Dir/Speed: 090 / 014 Kts Runway Surface: Concrete Temperature (°C): 32 Precip/Obscuration: No Obscuration; No Precipitation Runway Surface Condition: Dry

Pilot-in-Command

Age: 59

Certificate(s)/Rating(s)

Airline Transport; Flight Instructor; Multi-engine Land; Single-engine Land; Helicopter

Instrument Ratings Airplane Flight Time (Hours)

Total All Aircraft: 19500 Last 90 Days: 227 Total Make/Model: 7000 Total Instrument Time: UnK/Nr

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After an asymmetrical deployment of the thrust reversers during landing rollout deceleration, the captain failed to maintain directional control of the airplane and it veered off the runway, collapsing the nose gear and damaging the forward fuselage. Several days before the flight the #1 thrust reverser had been rendered inoperative and mechanically locked in the stowed position by maintenance personnel. In accordance with approved minimum equipment list (MEL) procedures, the airplane was allowed to continue in service with a conspicuous placard noting the inoperative status of the #1 reverser placed next to the engine's thrust lever. When this crew picked up the airplane at the departure airport, the inbound crew briefed the captain on the status of the #1 thrust reverser. The captain was the flying pilot for this leg of the flight and the airplane touched down on the centerline of the runway about 1,200 feet beyond its threshold. The captain moved both thrust levers into the reverse position and the airplane began yawing right. In an effort at maintaining directional control, the captain then moved the #1 thrust lever out of reverse and inadvertently moved it to the Take-Off/Go-Around (TOGA) position, while leaving the #2 thrust lever in the full reverse position. The thrust asymmetry created by the left engine at TOGA power with the right engine in full reverse greatly increased the right yaw forces, and they were not adequately compensated for by the crew's application of rudder and brake inputs. Upon veering off the side of the runway onto the dirt infield, the nose gear strut collapsed. The airplane slid to a stop in a nose down pitch attitude, about 7,650 feet from the threshold. There was no fire. Company procedures required the flying pilot (the captain) to give an approach and landing briefing to the nonflying pilot (first officer). The captain did not brief the first officer regarding the thrust reverser's MEL'd status, nor was he specifically required to do so by the company operations manual. Also, the first officer did not remind the captain of its status, nor was there a specific requirement to do so. The operations manual did state that the approach briefing should include, among other things, "the landing flap setting...target LAX02FA266 File No. 18288

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airspeed...autobrake level (if desired) consistent with runway length, desired stopping distance, and any special problems." The airline's crew resource management procedures tasked the nonflying pilot to be supportive of the flying pilot and backup his performance if pertinent items were omitted from the approach briefing. The maintenance, repair history, and functionality of various components associated with the airplane's directional control systems were evaluated, including the brake system, the nose landing gear strut and wheels, the brakes, the antiskid system, the thrust levers and reversers, and the throttle control unit. No discrepancies were found regarding these components.

### Brief of Accident (Continued)

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Occurrence #1: LOSS OF CONTROL - ON GROUND/WATER

Phase of Operation: LANDING - ROLL

## **Findings**

- 1. 1 ENGINE
- 2. THRUST REVERSER DISABLED
- 3. (C) THROTTLE/POWER CONTROL INADVERTENT ACTIVATION PILOT IN COMMAND
- 4. (C) DIRECTIONAL CONTROL NOT MAINTAINED PILOT IN COMMAND
- 5. (C) GROUND LOOP/SWERVE NOT CORRECTED FLIGHTCREW
- 6. (F) CREW/GROUP COORDINATION INADEQUATE FLIGHTCREW

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Occurrence #2: ON GROUND/WATER ENCOUNTER WITH TERRAIN/WATER

Phase of Operation: LANDING - ROLL

## **Findings**

7. TERRAIN CONDITION - GROUND

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Occurrence #3: NOSE GEAR COLLAPSED

Phase of Operation: LANDING - ROLL

#### **Findings**

- 8. LANDING GEAR, NOSE GEAR STRUT OVERLOAD
- 9. LANDING GEAR, NOSE GEAR STRUT FAILURE, TOTAL

Findings Legend: (C) = Cause, (F) = Factor

The National Transportation Safety Board determines the probable cause(s) of this accident as follows.

The captain's failure to maintain directional control and his inadvertent application of asymmetrical engine thrust while attempting to move the #1 thrust lever out of reverse. A factor in the accident was the crew's inadequate coordination and crew resource management.